



Thermal Challenges of Complex FPGA/3D ICs for Space Applications

by

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SEMI-THERM, March 18, 2013



Outline

- Package/Thermal
 - Package trend
 - Power density trend
- Thermal Challenges
 - General
 - Space
 - TM (Thermal Management) approaches
- Thermal Management
 - Low/high I/O CGAs
 - 3D ICs
- Summary



MPF to Curiosity (1996-2012)



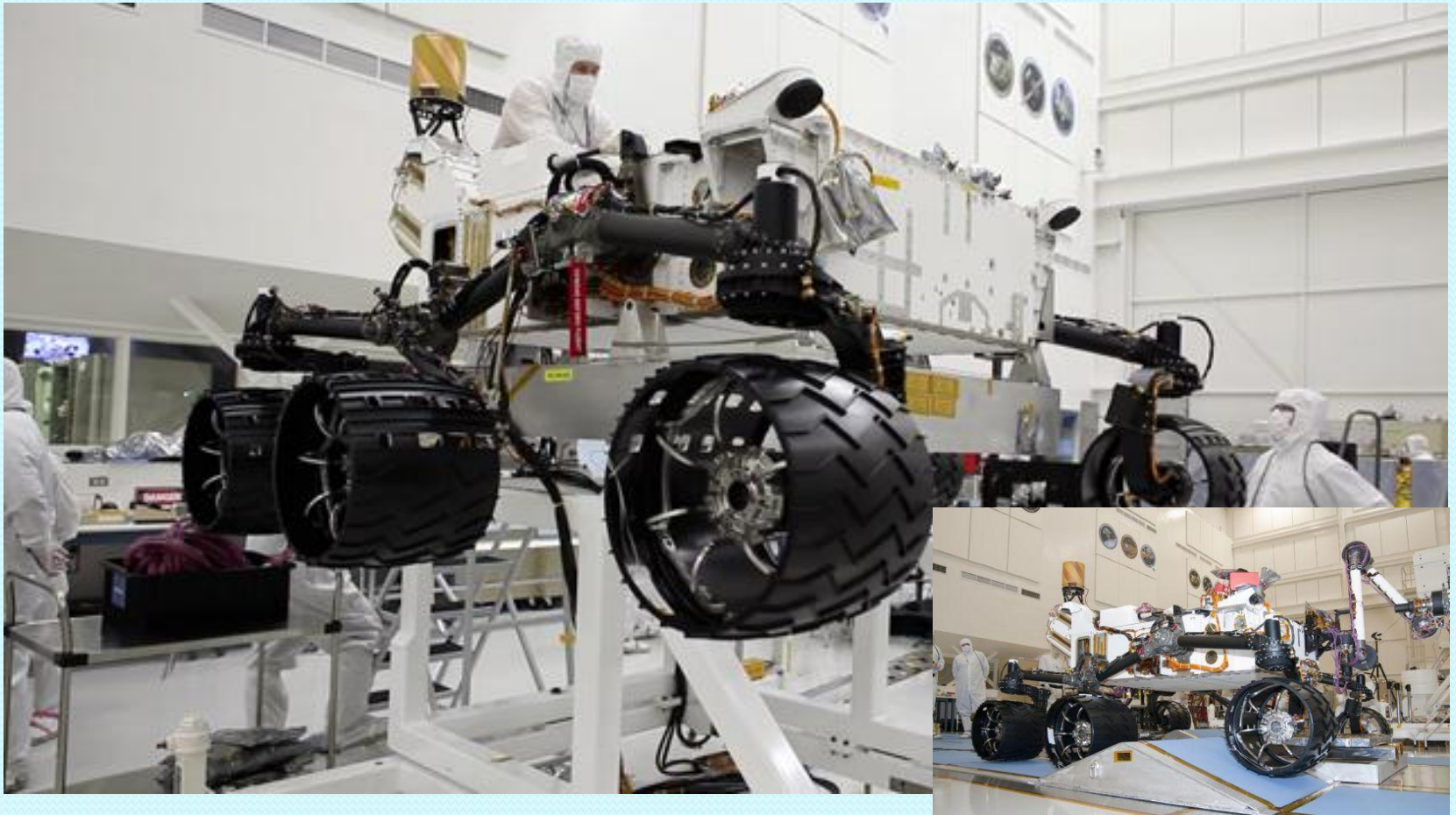


Curiosity Rover





Curiosity is Getting Ready for Mars



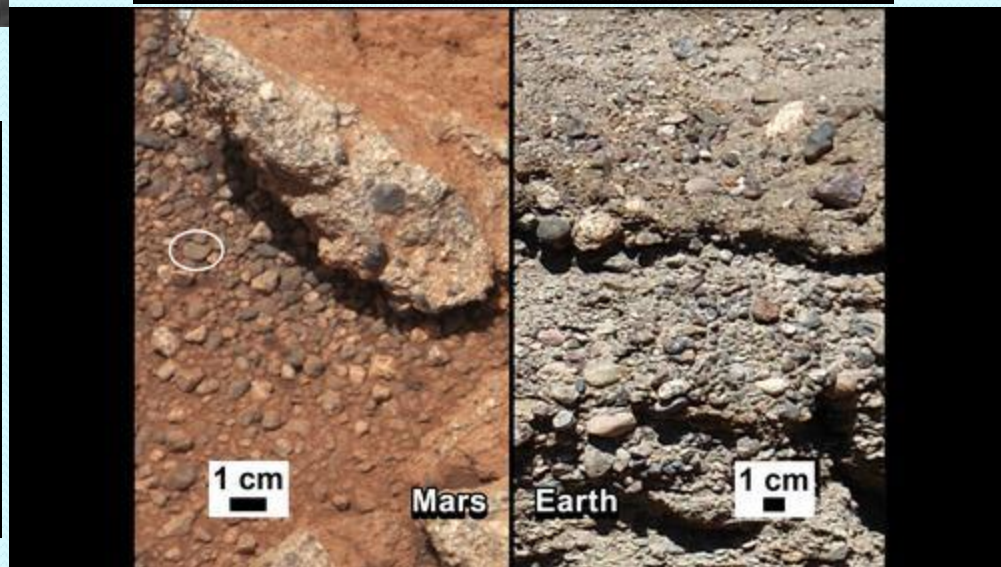
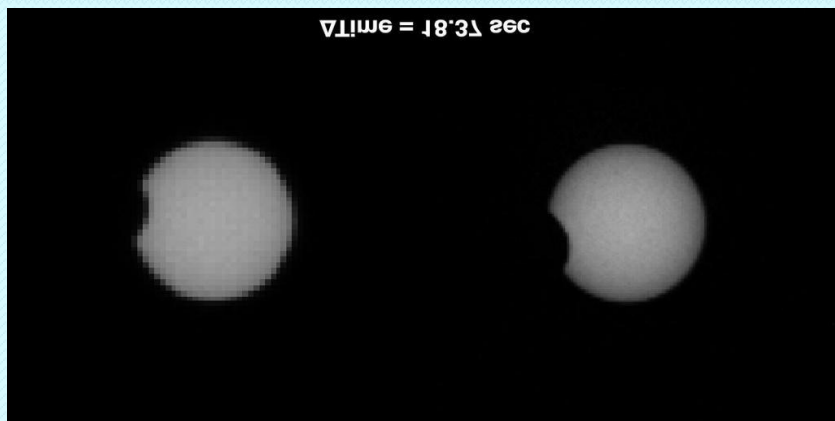
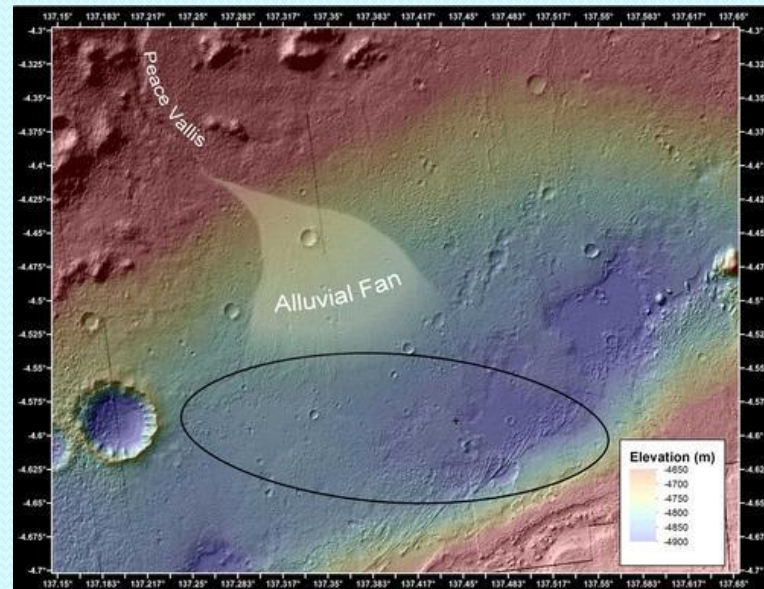
<http://spacespin.org/article.php/101043-watch-construction-curiosity-live>

Continuous live video of rover construction is available at:

<http://www.ustream.tv/channel/nasajpl>



Safe Arrival & Images of Mars

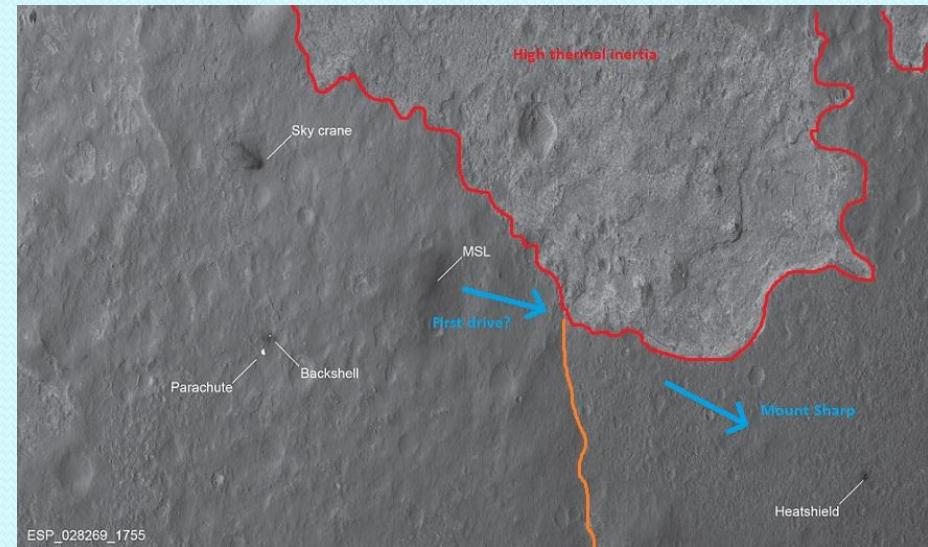
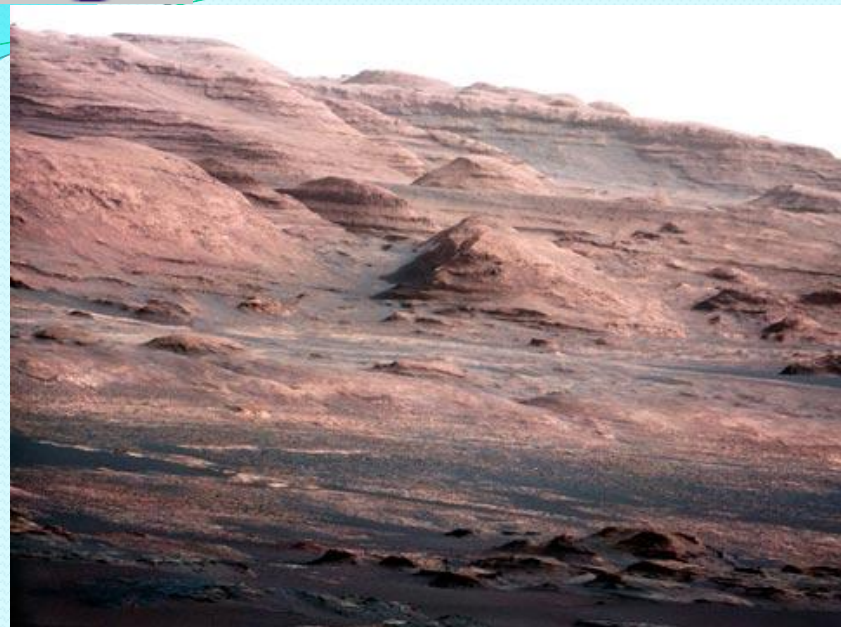




Images & 1st Time Drill on Mars/Planet

*Detected Chloro-methane which could be resulted from organic-decomposition
Such benign, could drink water*

NASA Briefing- Mar 12, 2013



ESP_028269_1755

SEMI-THERM , Mar. 18, 2013

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Curiosity Finds Habitable Env.



Opportunity- Sulfate rich sand stone
In presence of water
Not Habitable

Curiosity- Neutral pH
Habitable Env.
If microbes ever present



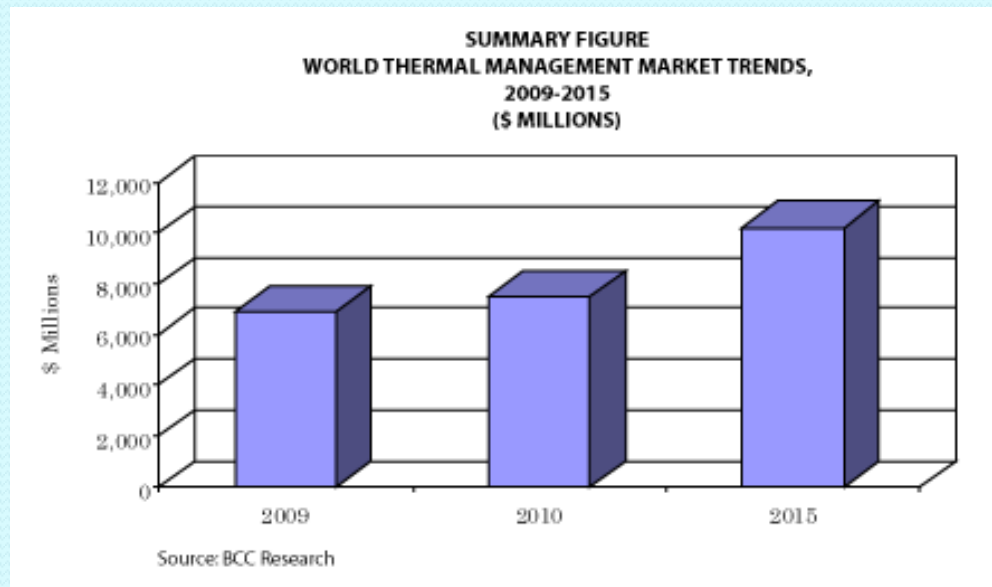
Why TM

Growth Market- \$10.2B in 2015

The North American

market will maintain its number one position throughout the period, with a market share of

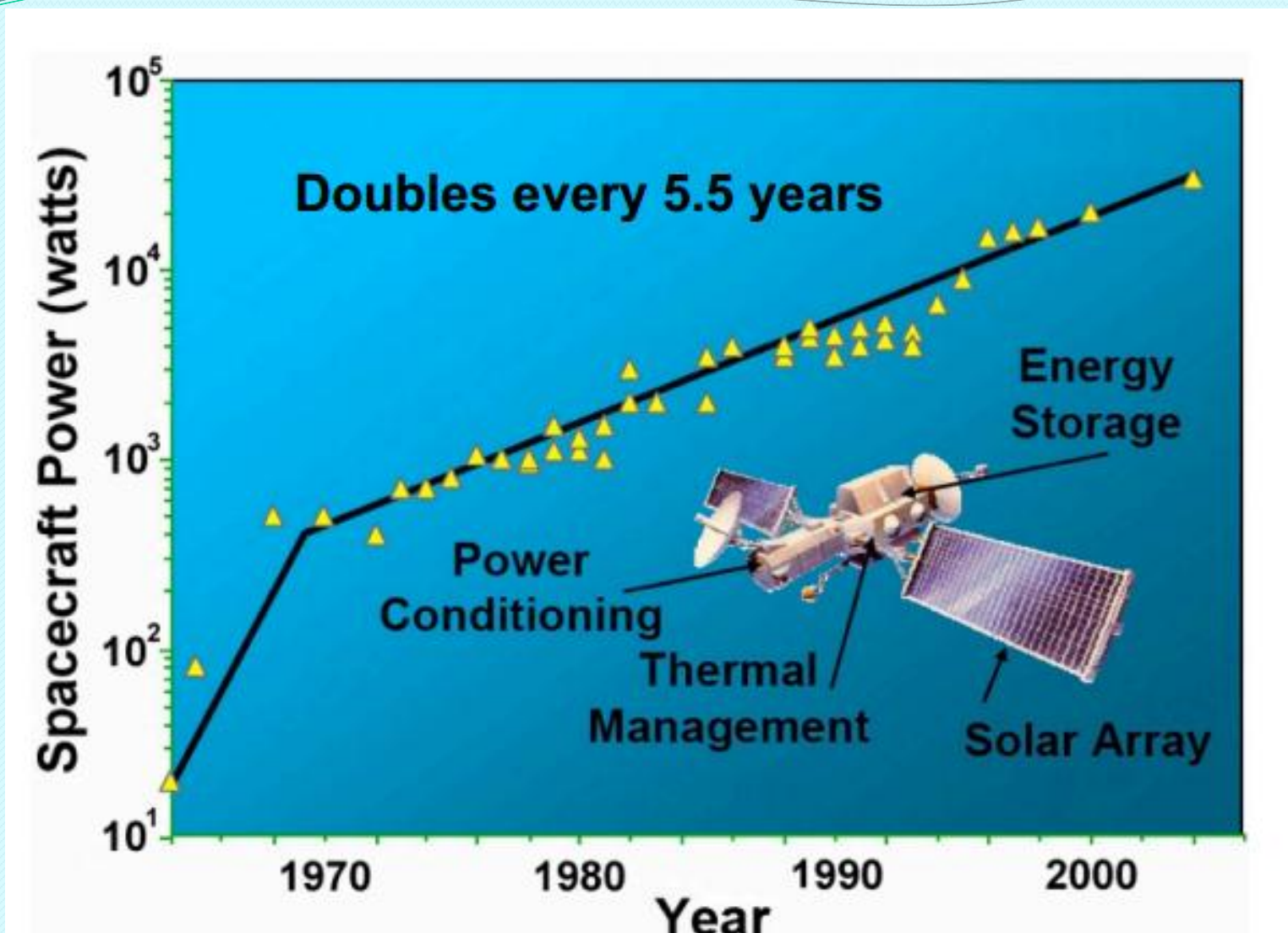
about **37%**, followed by **Asia-Pacific** with around **23%** to **24%**



- The world market for TM products to \$10.2 billion by 2015, a compound annual growth rate (CAGR) of 6.4% between 2010 and 2015.
- Asia-Pacific is projected to be the fastest-growing region, with a CAGR of 7.3% between 2010 and 2015. This region is valued at \$1.7 billion in 2010 and should reach \$2.5 billion by 2015.
- The market in the Americas is projected to grow from \$2.9 billion in revenues in 2010 to \$4 billion in 2015, a CAGR of 6.7%.

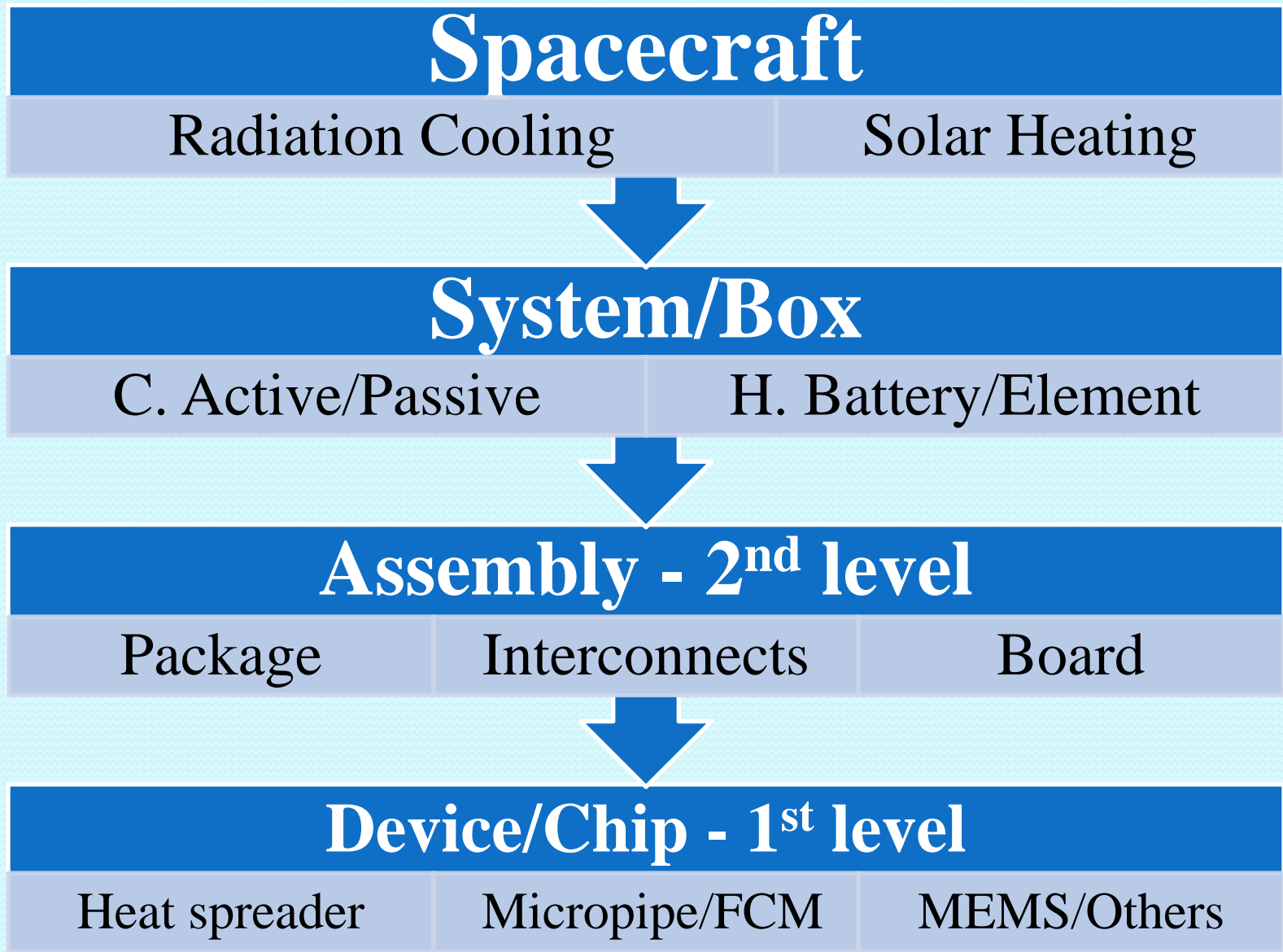


TM Must for Rising Power





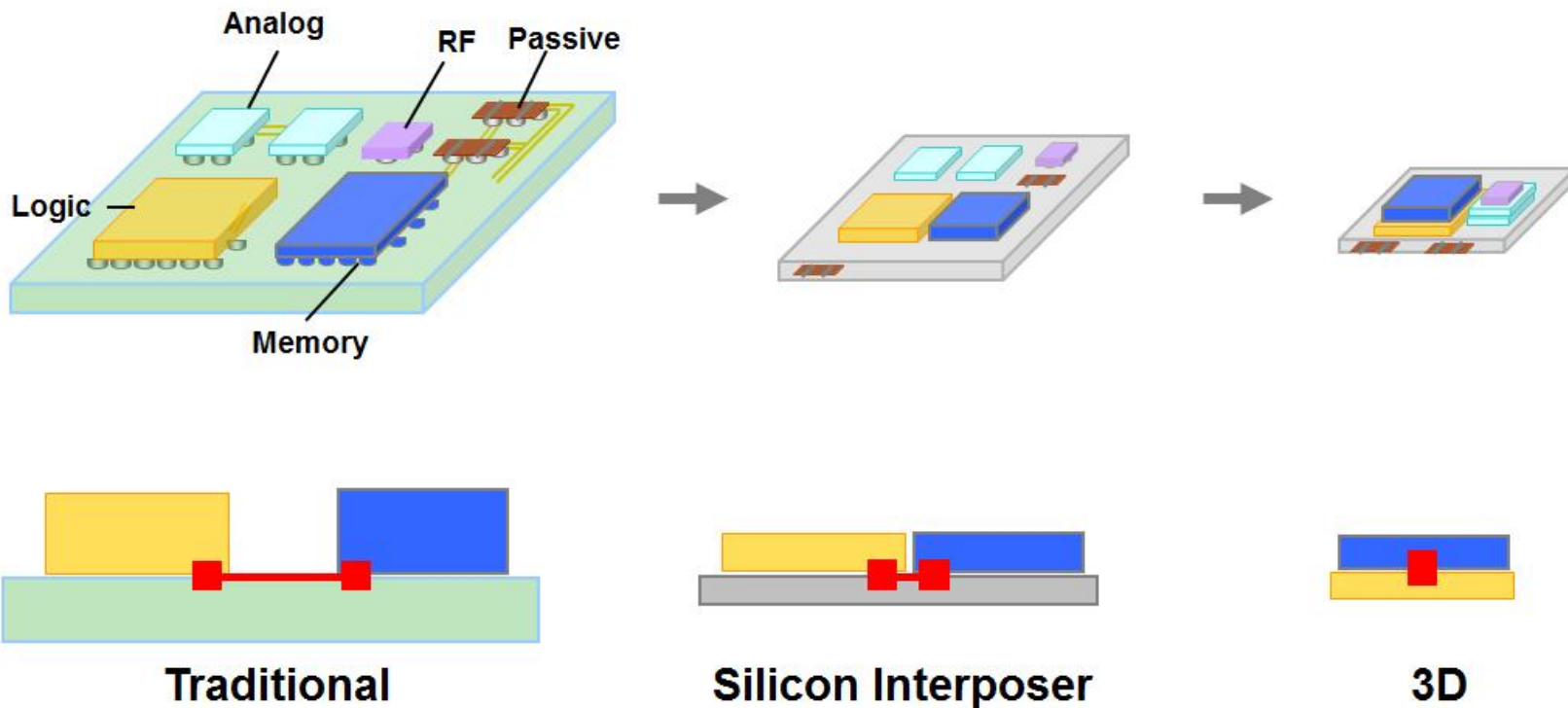
TM from System to Device





Density Increase Approaches

System Migration Trend

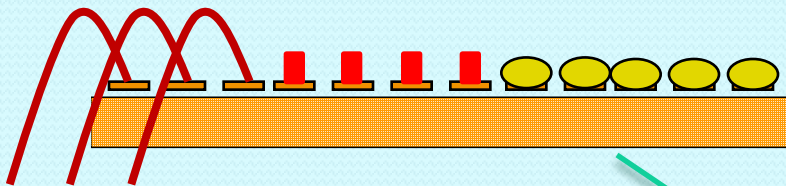


<http://www.i-micronews.com/news/Xilinx-brings-3D-TSV-interconnects-commercialization-phase,5693.html>

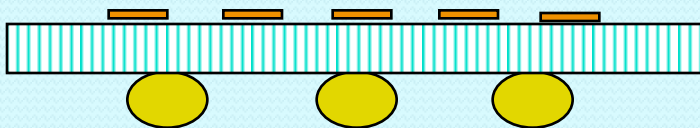


Advanced Package Concepts

- Die Tight Pitch
- Al Pad- Non Reflow

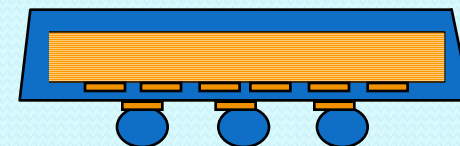


- Interposer
- Polymer, Ceramic, Flex
- Cu:Ni:Au Pad

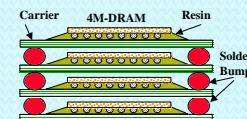


- Norm Pitch for PWB
- 0.3-1.27 mm

- Wafer
- Pitch limitation

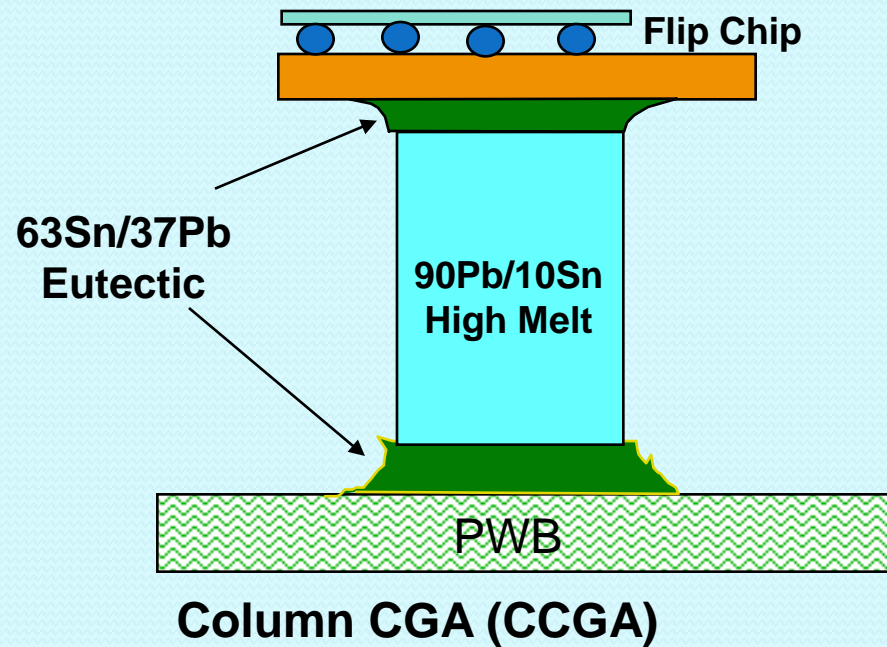
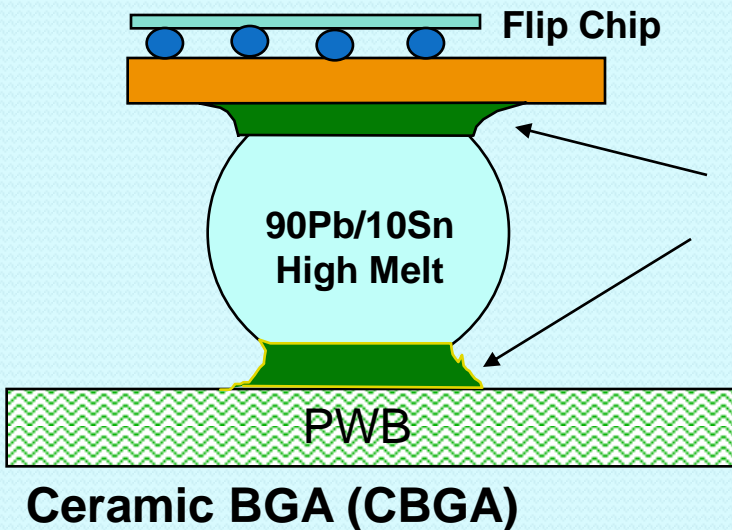
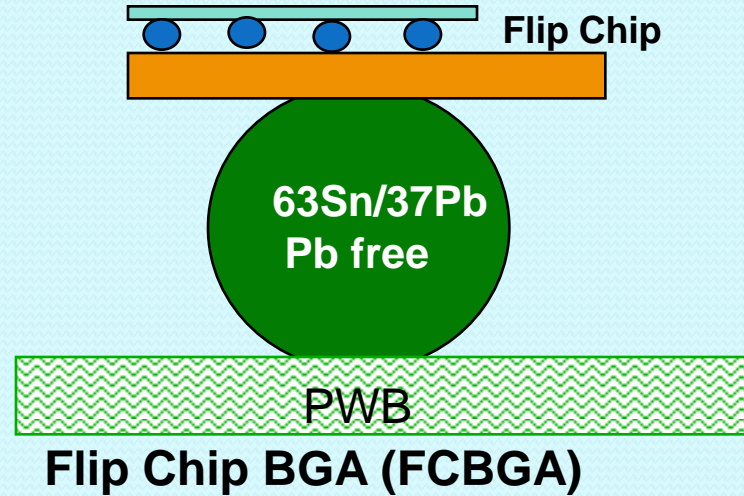
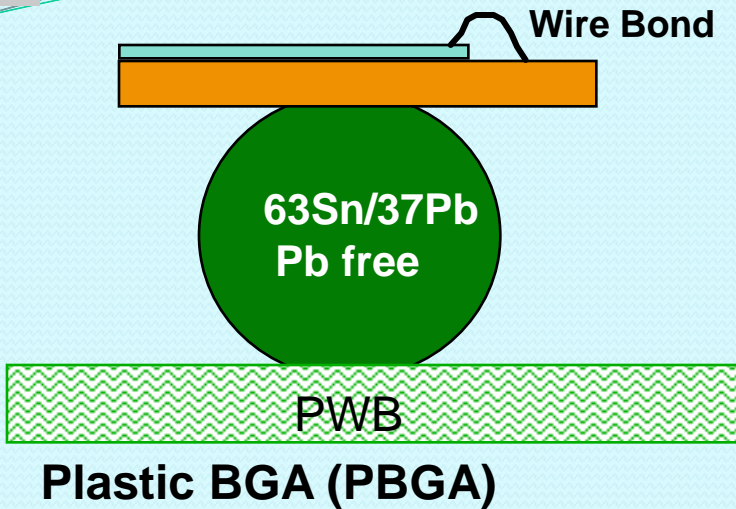


- Stack
- die/wafer/package



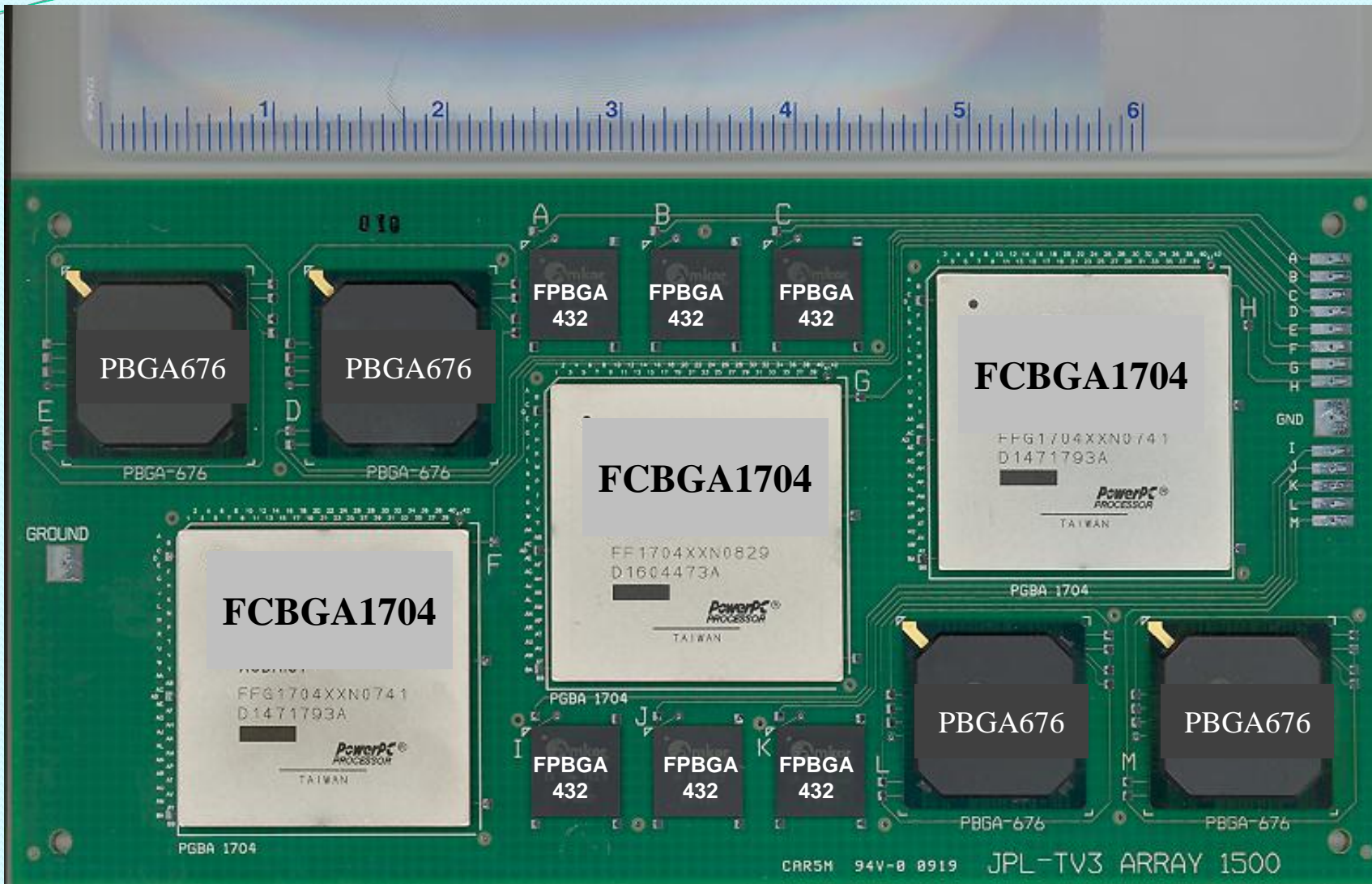


FCBGA and CGA



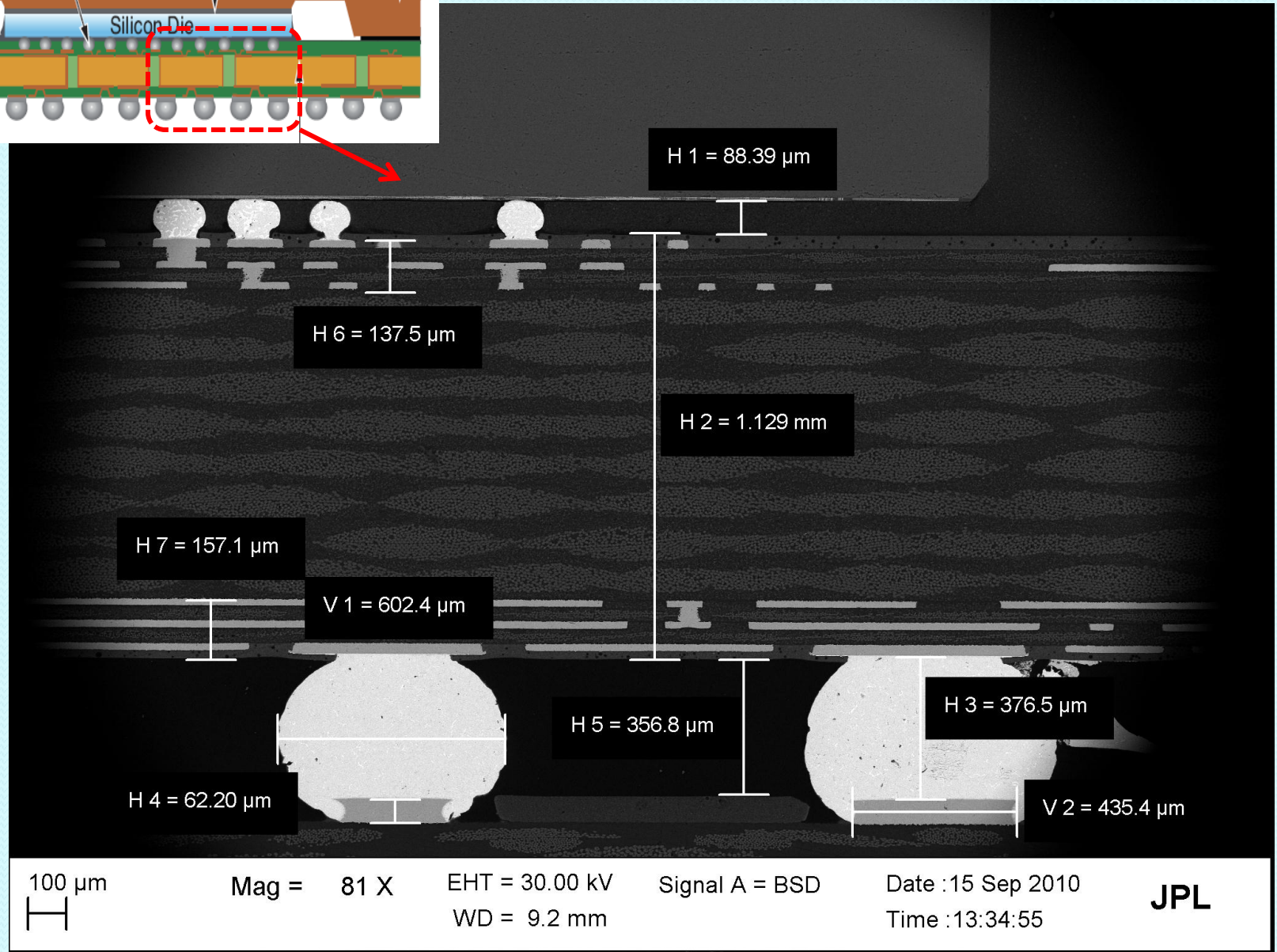
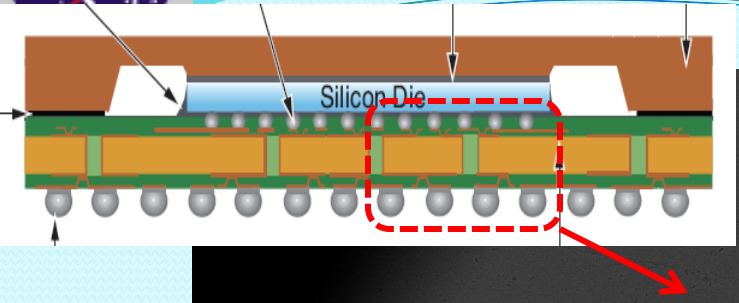


FCBGA , PBGA, FPBGA



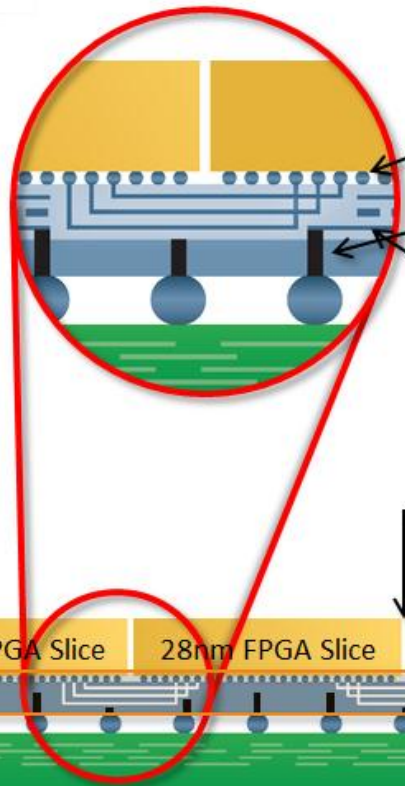


FCBGA Heat Sink & Build





Interposer/TSV & Dies



Microbumps

- Access to power / ground / IOs
- Access to logic regions

Through-silicon Vias (TSV)

- Only bridge power / ground / IOs to C4 bumps
- Coarse pitch, low density aids manufacturability
- Etch process (not laser drilled)

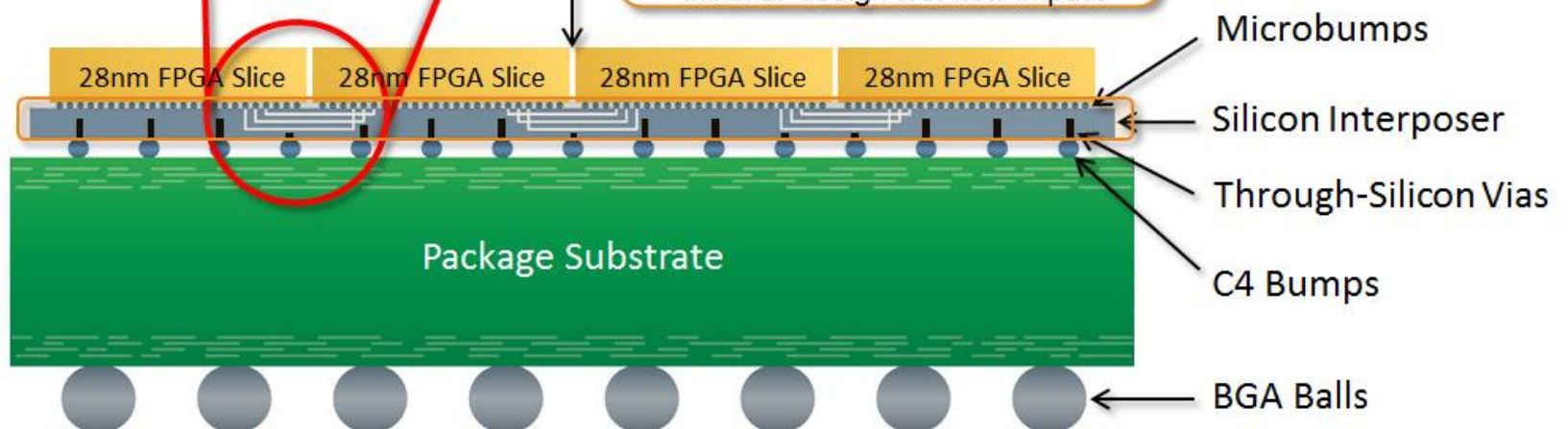
Passive Silicon Interposer (65nm Generation)

- 4 conventional metal layers connect micro bumps & TSVs
- No transistors means low risk and no TSV induced performance degradation

Side-by-Side Die Layout

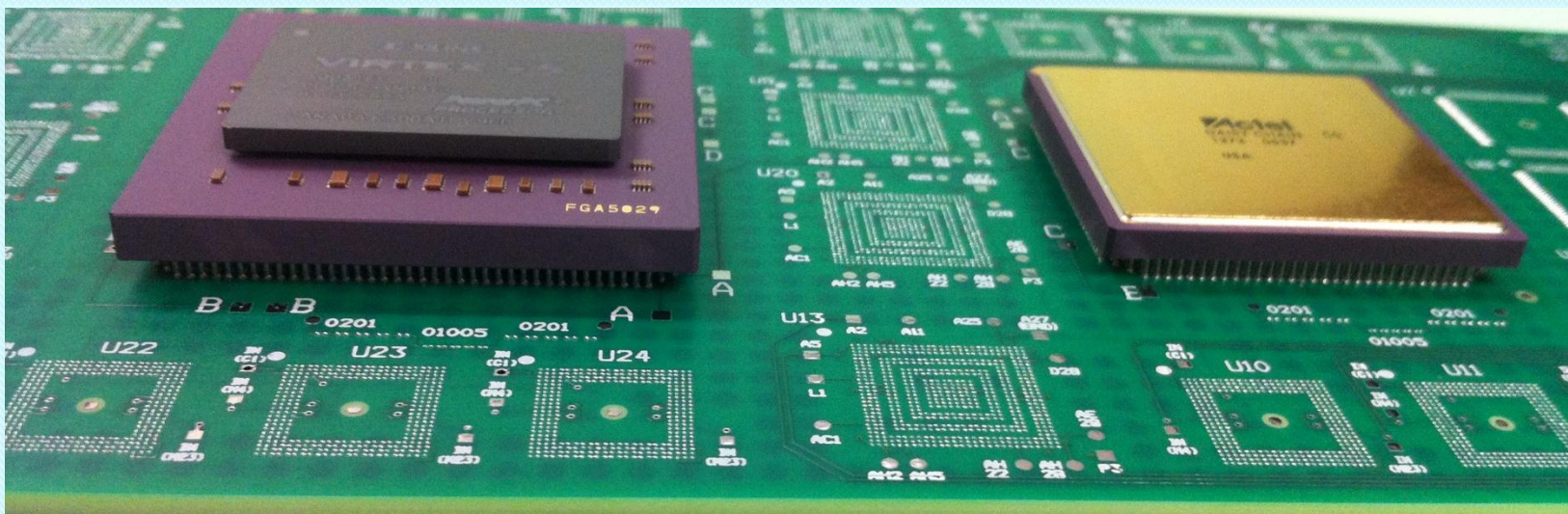
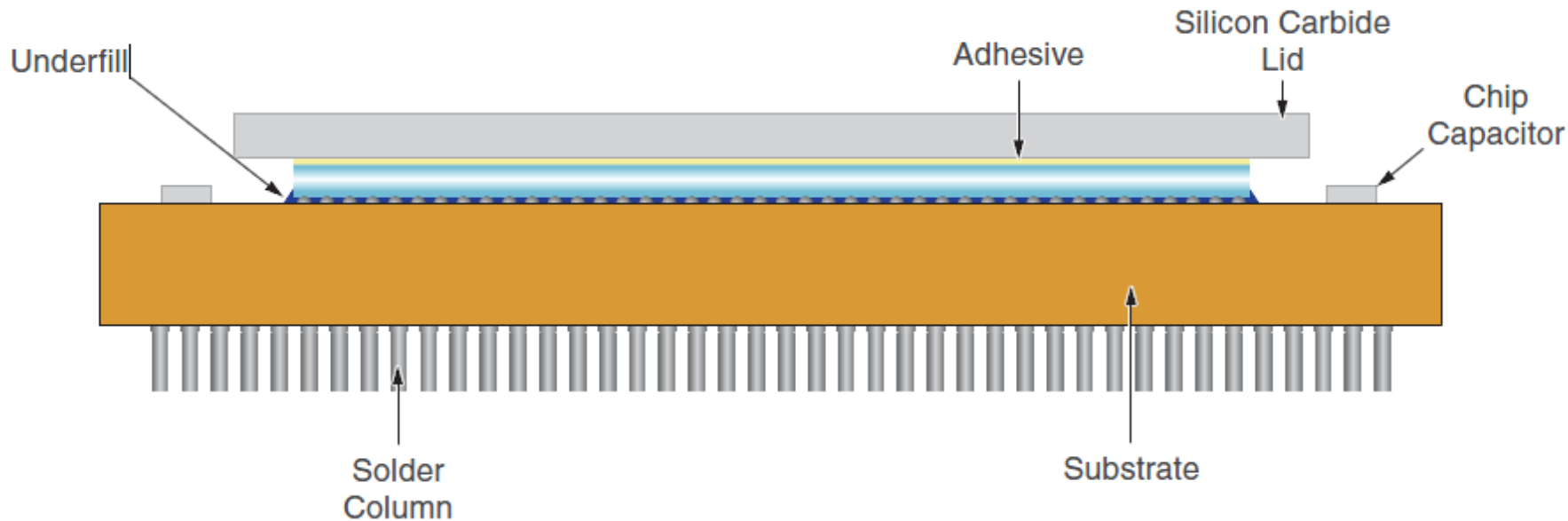
- Minimal heat flux issues
- Minimal design tool flow impact

New!



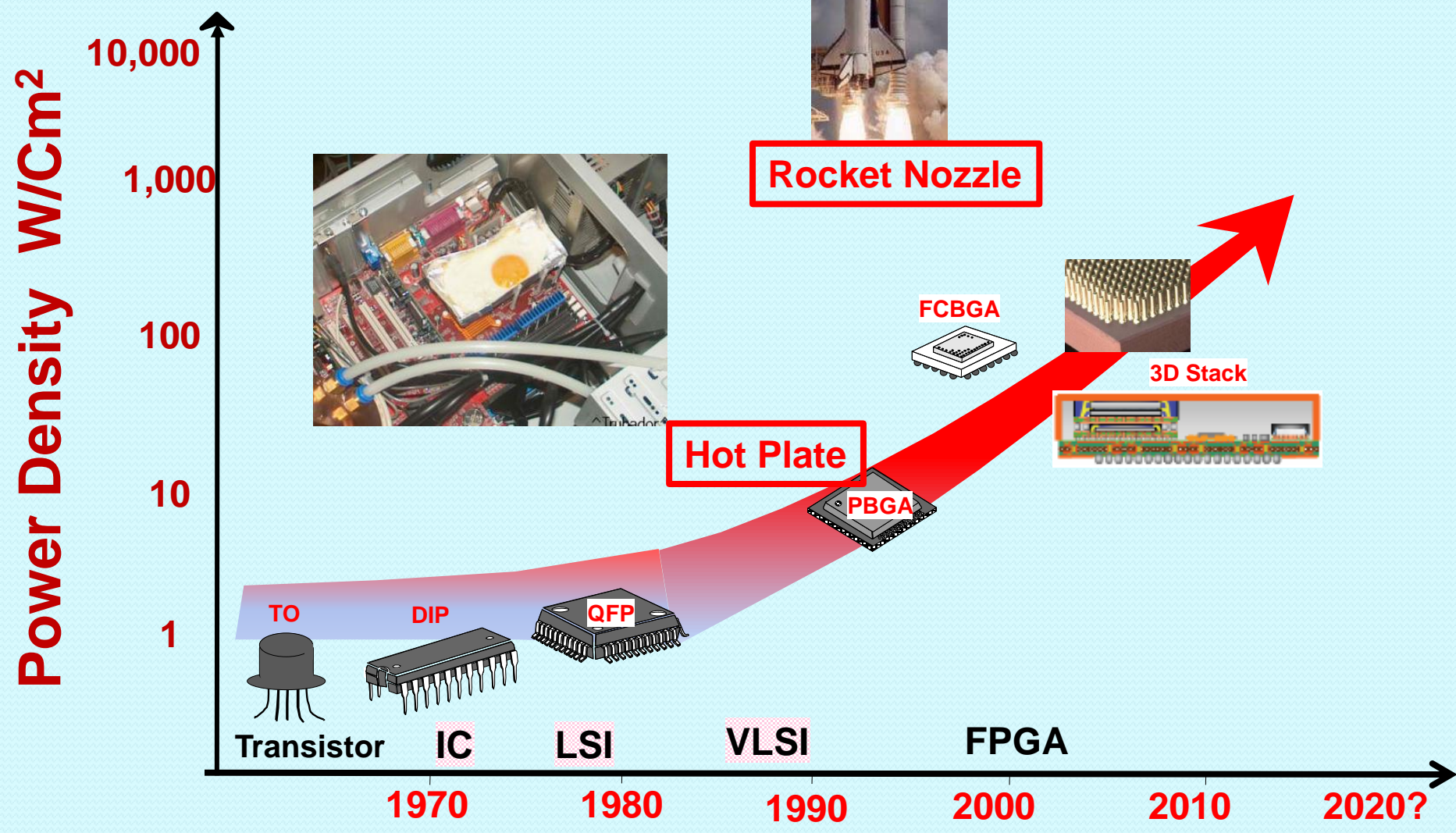


High I/O CGAs





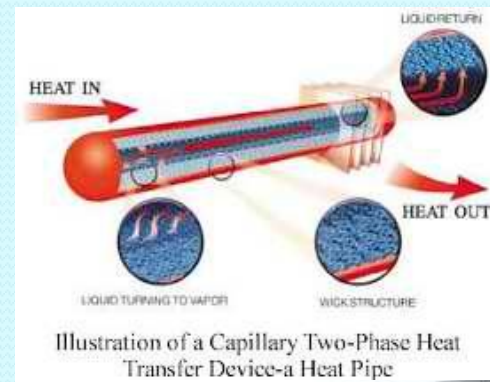
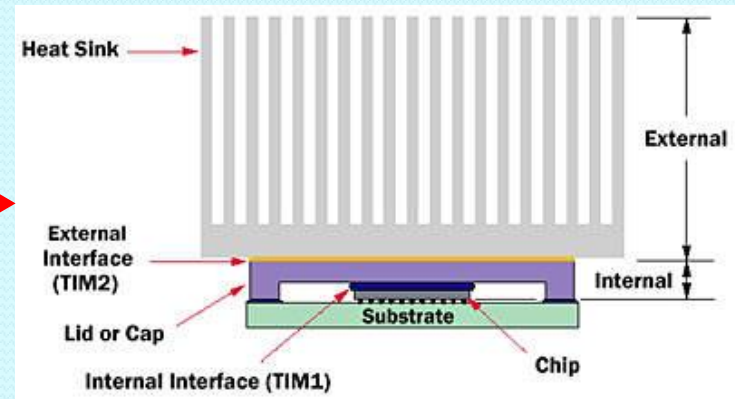
Power Density Explosion





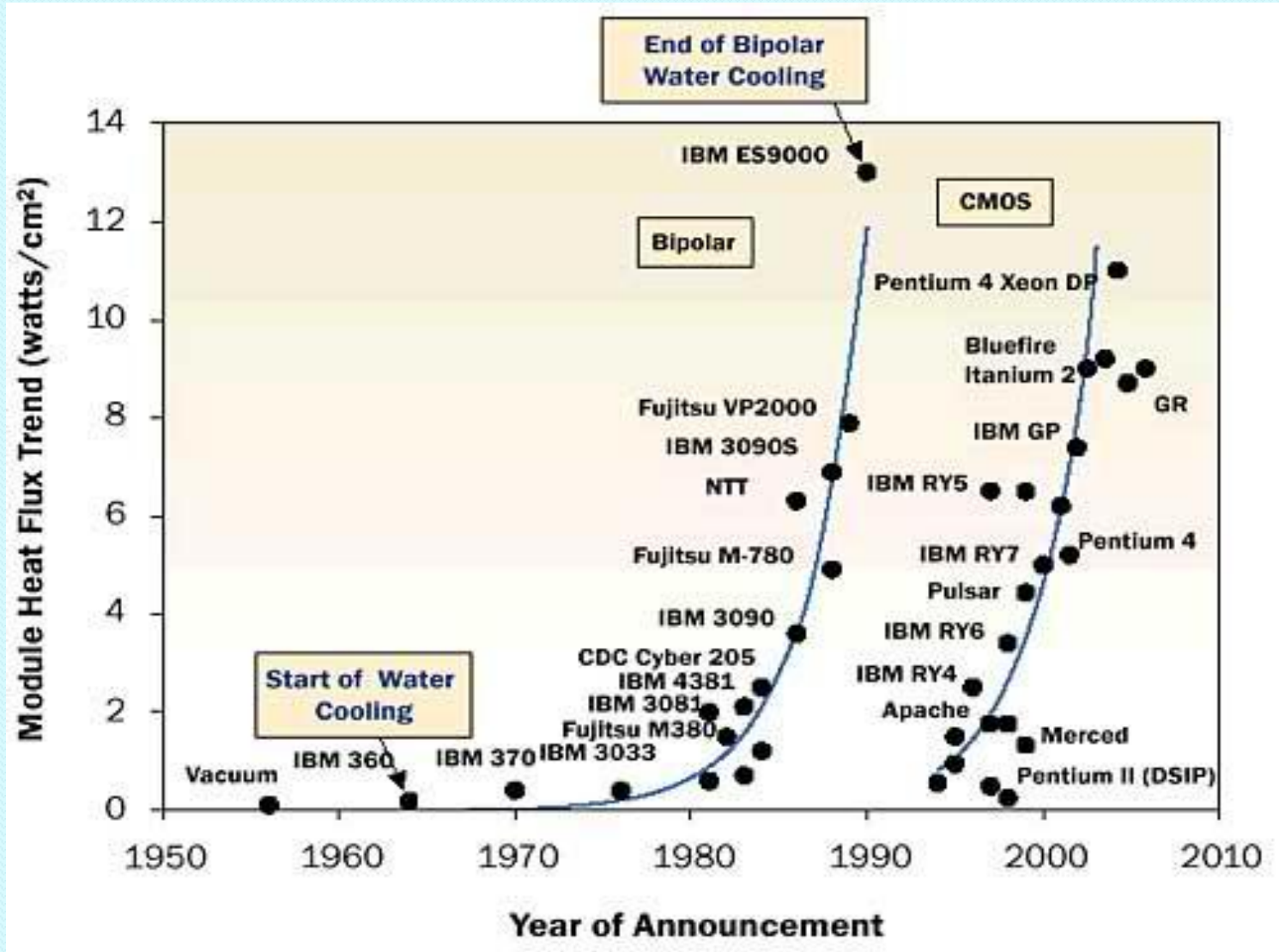
Cooling of Electronics

- Heat Spreading
- Air Cooling
 - Piezo fans
 - Synthetic jet cooling
- Liquid Cooling
 - Heat pipes
 - Cold plates
 - Micro-channels
- Liquid Metal/Immersion
- Solid State/Spray Cooling
- More ...





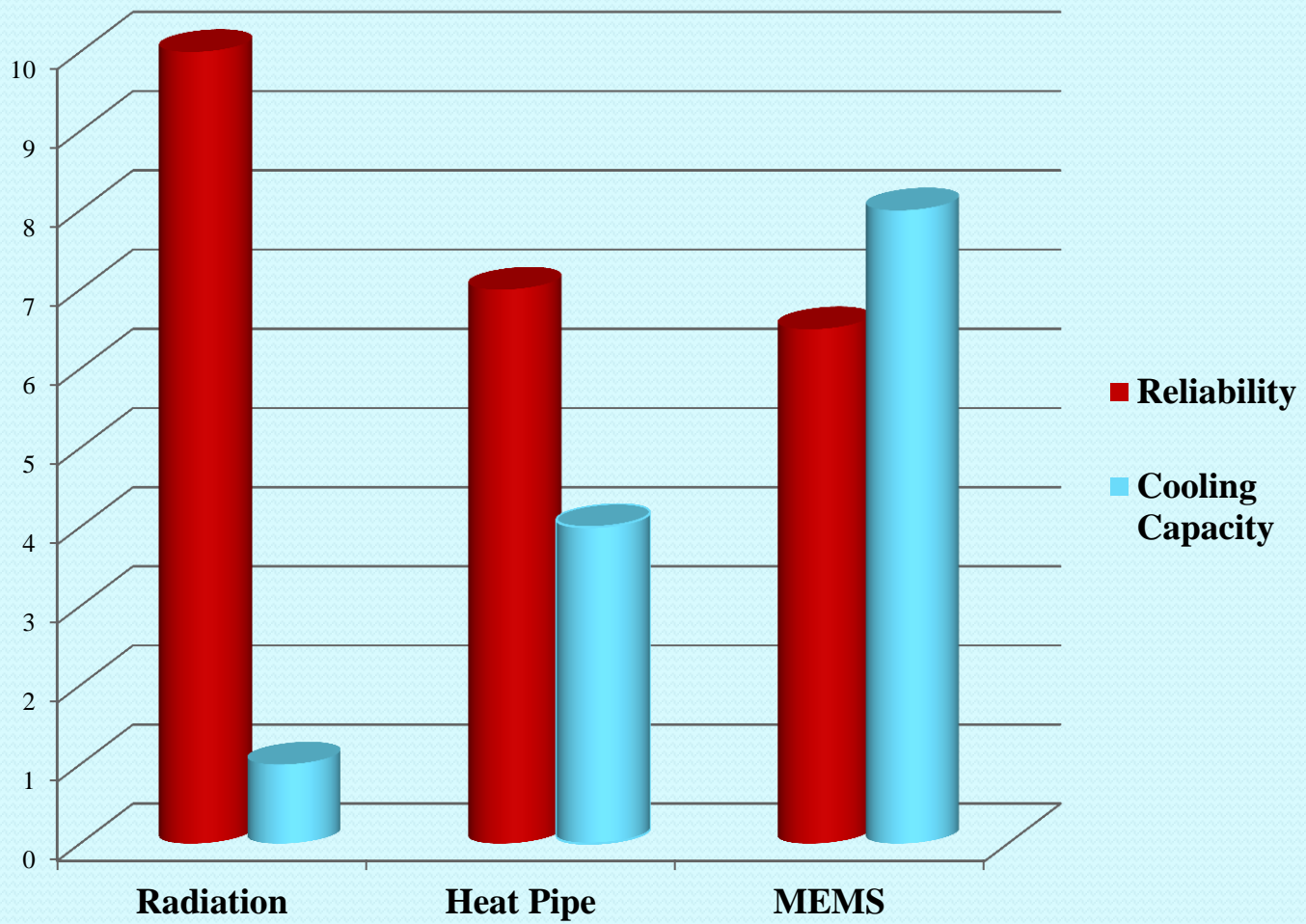
Power & Encore of Water Cooling



<http://www.electronics-cooling.com/2005/08/liquid-cooling-is-back/>



TM Techniques & Reliability

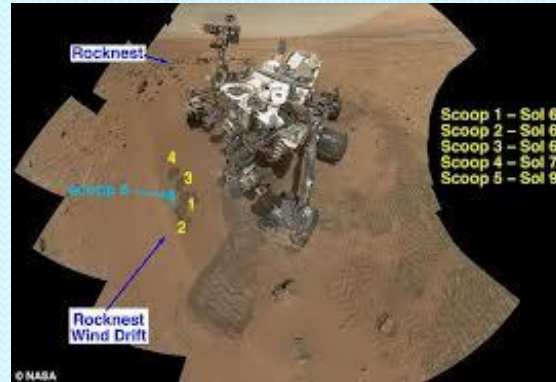




TM Earth vs. Space

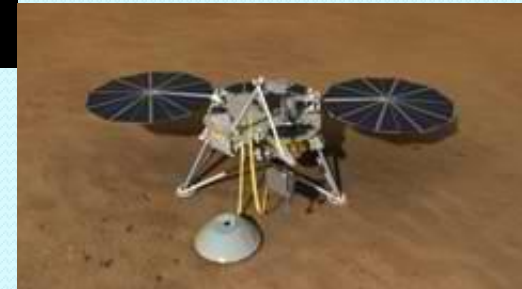
- Air of Earth

- Radiation
- Convection
- Conduction



- Vacuum of Space

- Radiation to space
- Controls (passive/active), coating/MLI/Louver



- Why TM

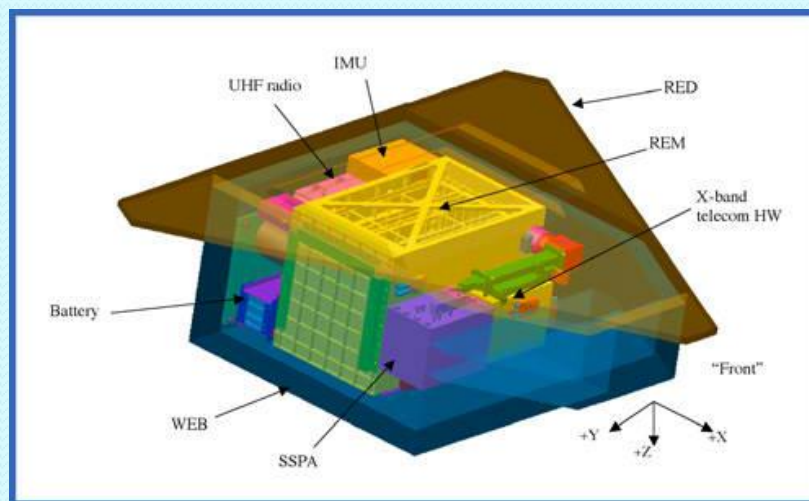
- Control Min/Max Temp. range for electronics use
- Control Temp. range for electronics/materials stability
 - Use solar/RTG (Radio-isotope Thermoelectric Generator)/battery
 - Mars Rovers, e.g., W/WO env. Controls WEB/camera



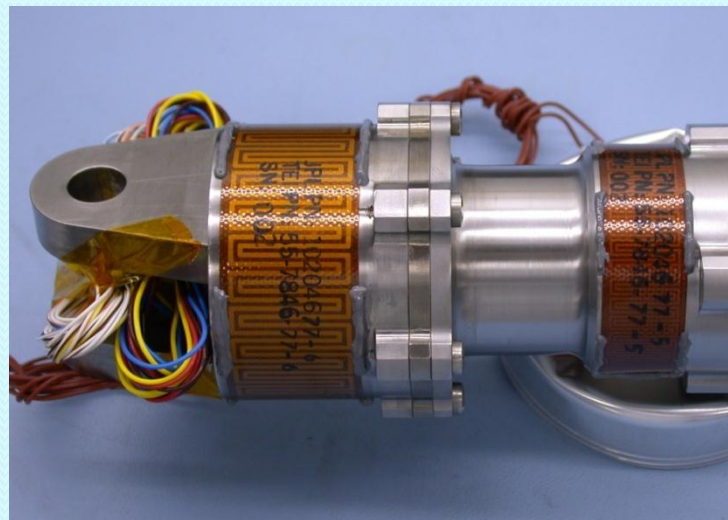
Heater to Keep Electronics Warm

W

- Control hardware/electronics Temp.
 - Global control, Warm Electronic Box (WEB)



- Local heaters ($P = I^2R$)
 - Kapton film heaters
 - Wire wound resistors
 - Cartridge heaters
 - Radioactive heating units
 - Others





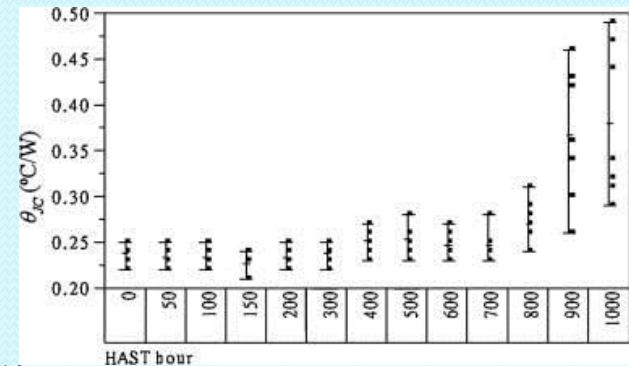
Space Cooling Technologies

- Passive cooling
 - Rolling it, sun side cools off
 - Coating (white/gold)/MLI/shield
 - Radiator, Louver shutter, heat pipe
 - **Simple/reliable/low cost, but cooling limited**
- Active cooling
 - Mechanically pumped loops
 - Recuperative cryocooler, Optical cryocooler, etc.
 - **Complex/less reliable/hi cost, cools to lower T**
- Stored cryogen
 - Superfluid helium, solid nitrogen,



TM Device/Board

- Device Level- heat spreaders
 - High performance thermal interfacial materials thermal grease, thermal paste and PCM
 - Low CTE, high thermal conductivity pkg. materials
 - **Heat pipe/pump loop**
 - Embedded micro-heat pipes
 - Micro- or mini-channel heat sink
- Board Level
 - Single phase micro-mini channel cooling
 - Phase change cooling, ultra thin-film evaporator cooling
- Software/Hardware
 - Static, offline at design level
 - Dynamic, online during execution level





Constrained Vapor Bubble (CVB)



PI: Prof. Peter C. Wayner, Jr., Rensselaer Polytechnic Institute
 Co-I: Prof. Joel L. Plawsky, Rensselaer Polytechnic Institute
 PS: David F. Chao, NASA GRC
 PM: Ronald Sicker, NASA GRC
 Engineering Team: ZIN Technologies, Inc.

Glenn Research Center

Objective:

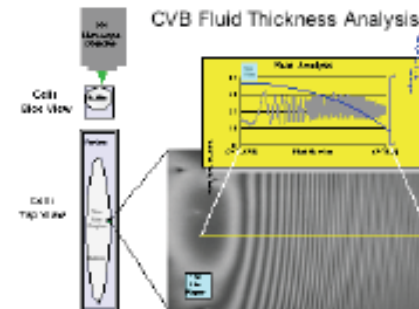
- To determine the overall stability, the fluid flow characteristics, the average heat transfer coefficient in the evaporator, and heat conductance of the constrained vapor bubble, under microgravity conditions, as a function of vapor volume and heat flow rate.

Relevance/Impact:

- CVB is crucial for engineering heat pipes for space applications.
- CVB flow induced by capillary flow eliminating need for wicks.
- Data from CVB will lead to optimally designed heat pipes that will operate at full capacity and provide large weight savings.
- CVB will provide the understanding of the maximum achievable performance of simple heat pipes based on corner flows.

Development Approach:

- The CVB/LMM is designed for autonomous operation through scripts and ground commanding. Crew time is required for initial installation and check out in the Fluids Integrated Rack (FIR), sample change out, and removal from FIR.
- The LMM and CVB flight hardware was developed under a proto-flight approach with the exception of the CVB module which follows the traditional qual/flight approach. The CVB modules will have spares, all other spare hardware will be kitted and assembled as required.
- The LMM and CVB are designed to utilize the FIR capabilities to the maximum extent possible.



ISS Resource Requirements

Accommodation (carrier)	Fluids Integrated Rack (FIR)/LMM
Upmass (kg) (w/o packing factor)	58 Kg for CVB
Volume (m ³) (w/o packing factor)	0.025 CVB
Power (kw) (peak)	0.5kw for CVB/LMM 1.1 kw for FIR/CVB/LMM
Crew Time (hrs) (Installation/operations)	34 Hours
Autonomous Operations	2wks/module 5 modules = 10 wks
Launch/Increment	17A/Increment 19

Project Life Cycle Schedule

Milestones	SCR	RDR	PDR	CDR	VRR	Safety	FHA	Launch	Ops	Return	Final Report
Actual/ Baseline	9/97	12/98	2/02	12/03	8/04	Phase III 11/05	3/09	8/09	Inc. 19/20	2010	2011
Documentation	Website: eRoom:				SRD: EDMP:			Project Plan: SEMP:			

Revision Date: 9/22/2008



Heat Pipe Study on Space Station



Light Microscopy Module / Constrained Vapor Bubble

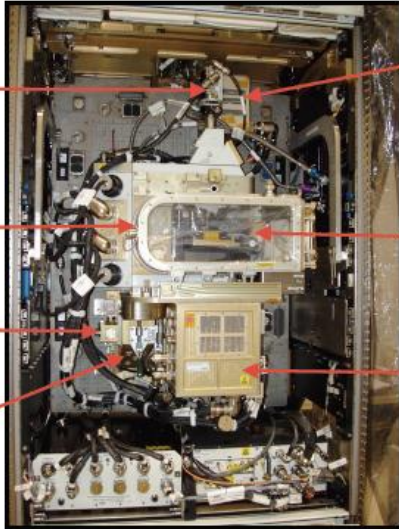
Glenn Research Center

QImaging
Camera

Microscope

SAMS TSH

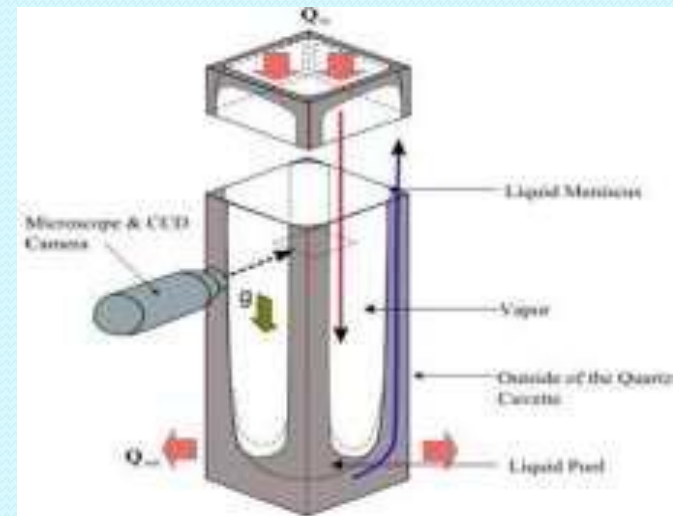
Spindle
Bracket



CVB Control Box

Auxiliary Fluids
Container

LMM Control Box



- **Cuvette heat pipe: Rectangular-shaped glass tube of quartz**
- **Observation of fluid flow**
- **Accurate temp. measurement**
- **Operate on earth, then at microgravity ISS (higher T/P)**
- **No convection to cool pipe surface, only radiation in Space**
- **Understand basic principles, heat transfer with no moving part**

http://www.nasa.gov/mission_pages/station/research/news/heat_pipes.html

SEMI-THERM , Mar. 18, 2013

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Heat-Pipe for Commercial Use



NASA heat pipe technology used in spacecraft to keep hardware and critical electronics cool has found its way into notebook computers. Tiny heat pipes are in wide use to cool the main central processor chip.

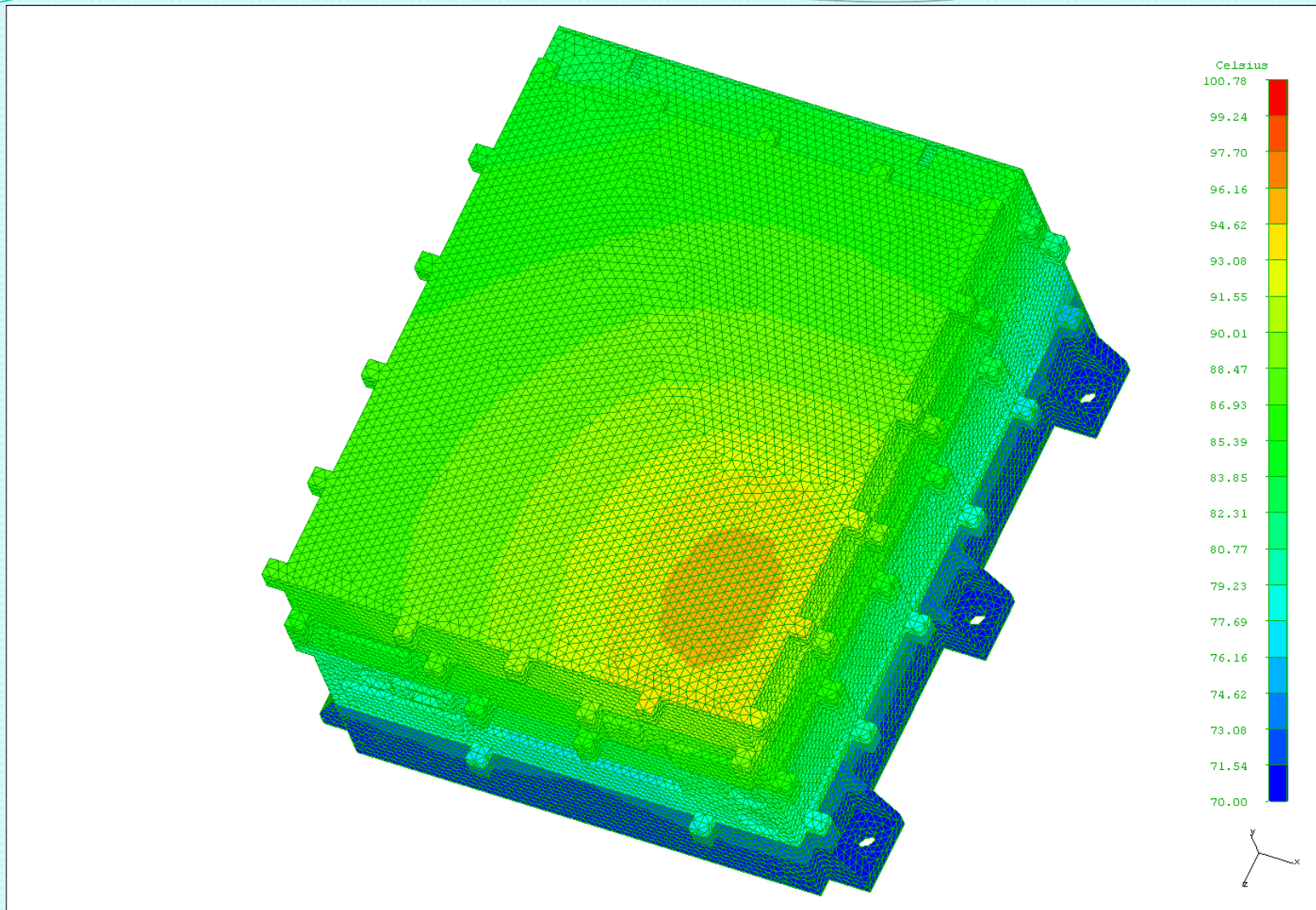
<http://spinoff.nasa.gov/spinoff1997/ct4.html>

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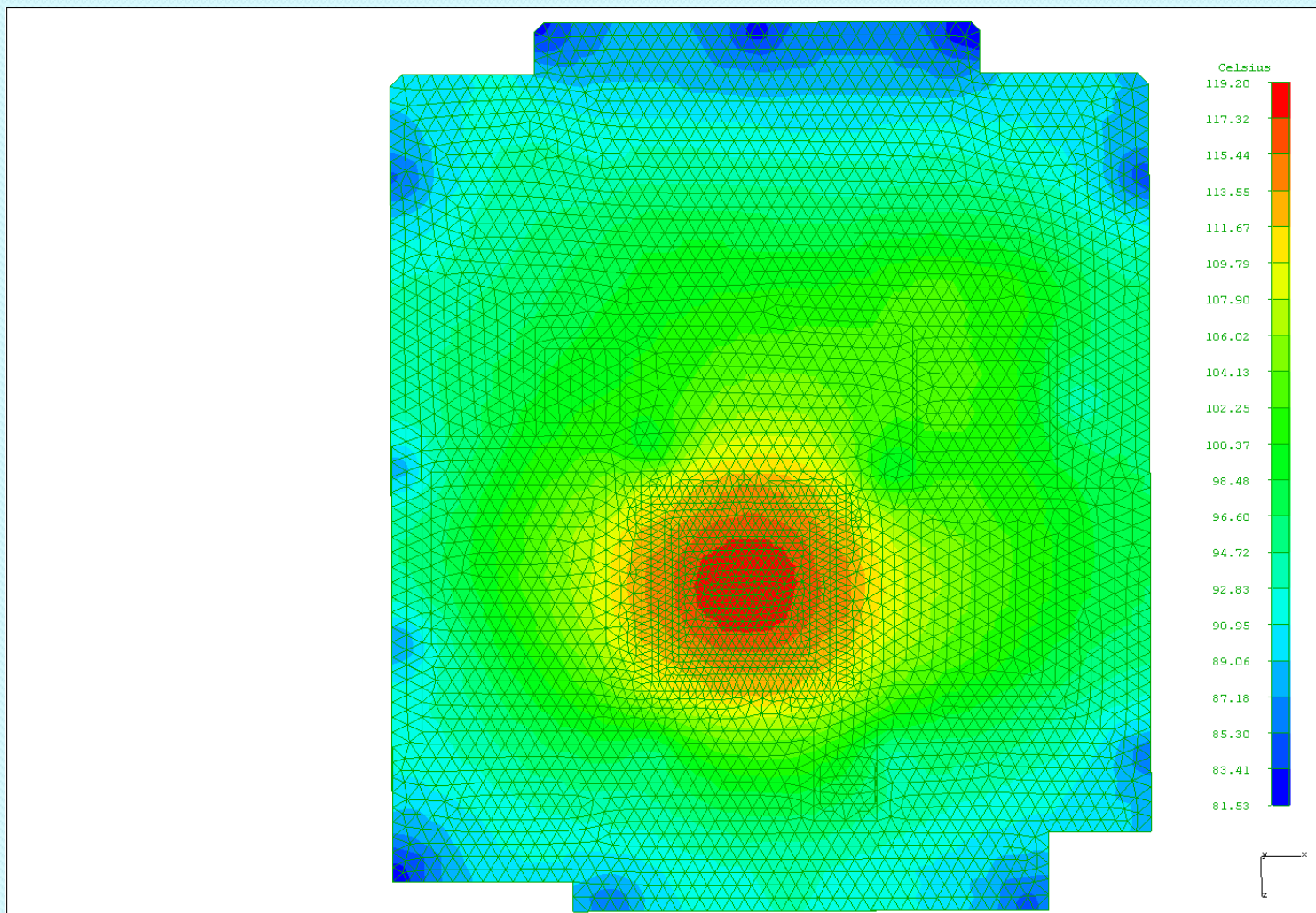
Box Temp Rise



Temperature Contour Plot – Low Power



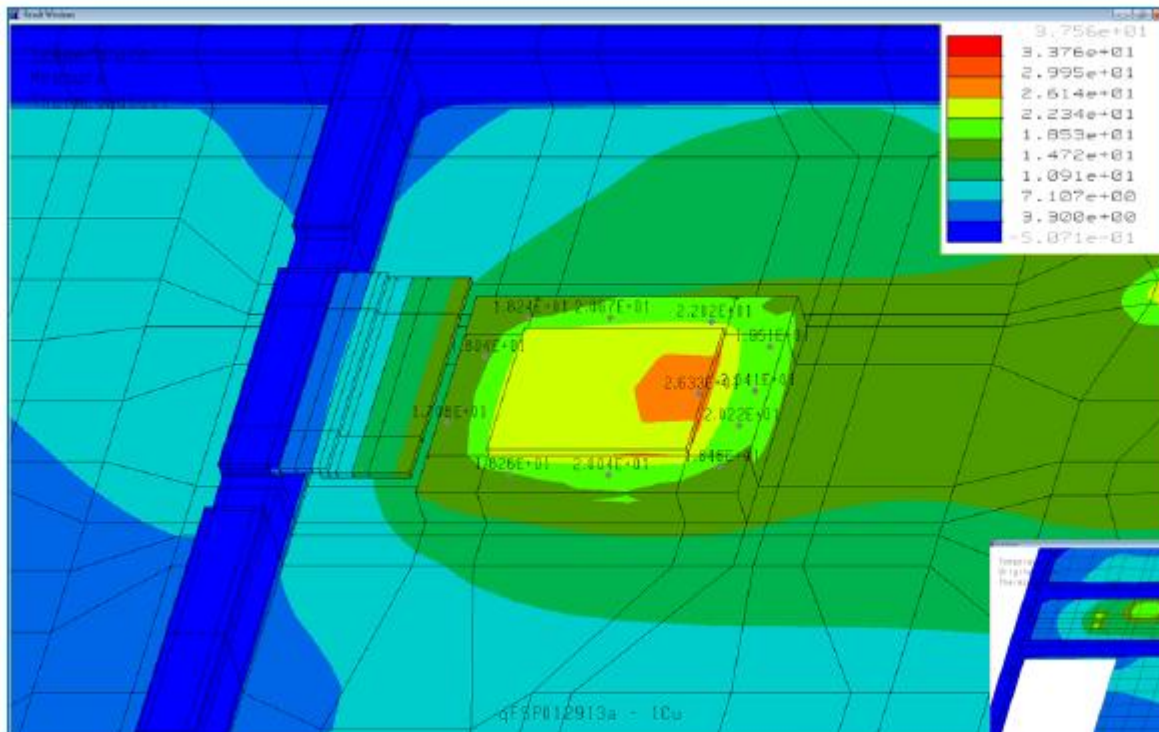
FPGA Temp Rise – High Power



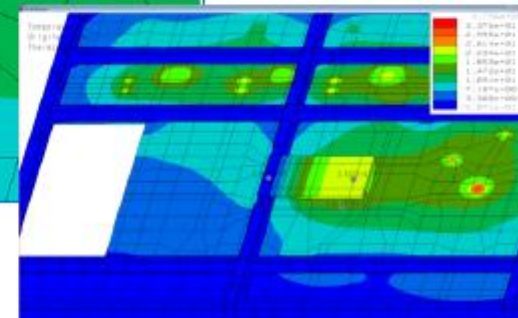
Total rise 49.2 °C from baseplate temp (70 °C B/P)



High I/O CGA Temp- WO/W Heat Sink



Heat Flows: 11.7W out of 15W goes thru Cu strap



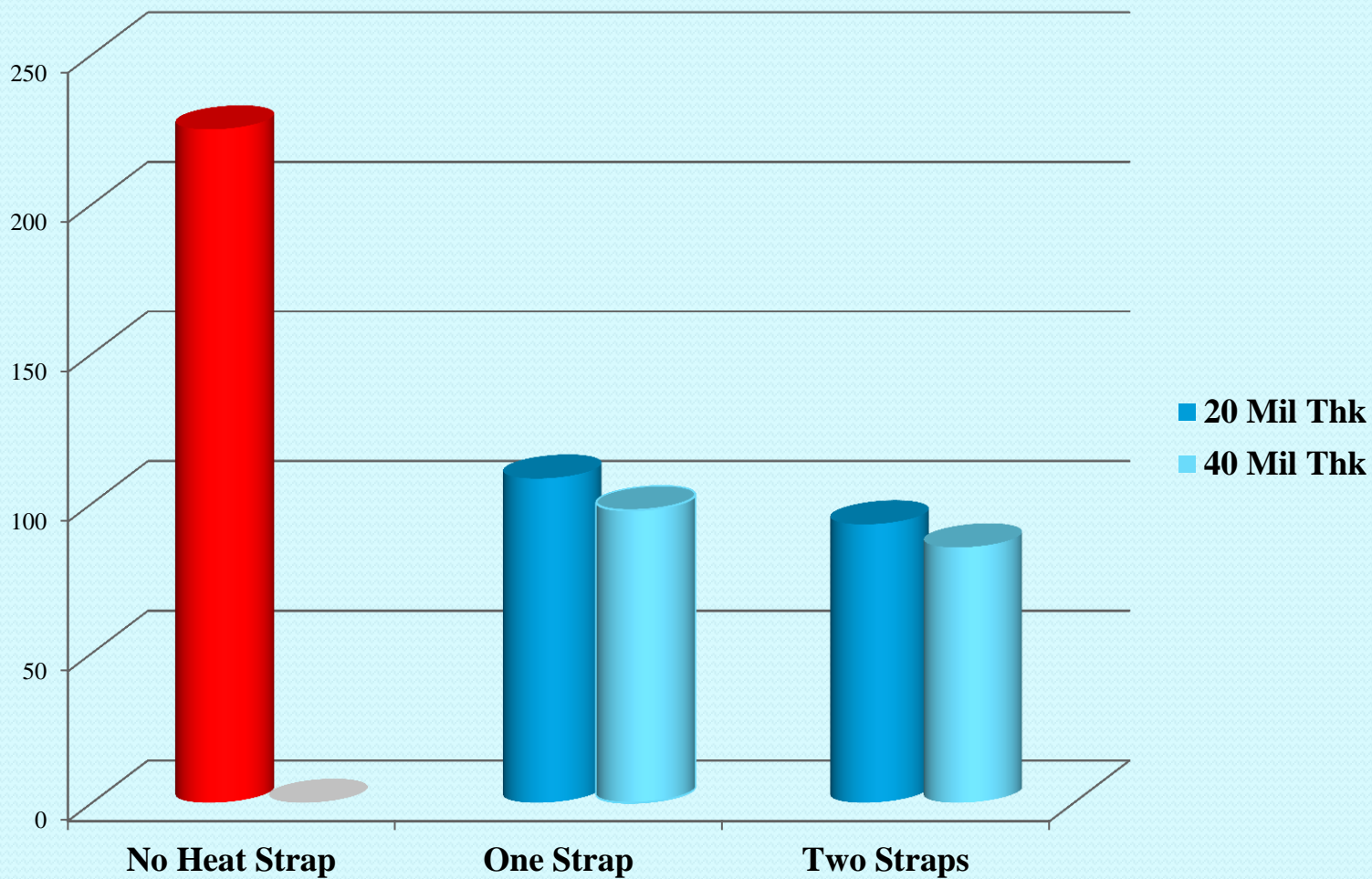
No Heat Sink-Max Temp 225 °C (70°C B/P)

With Heat Sink- Max Temp 98°C (70°C B/P)

Ref: S. Tseng JPL, 2013

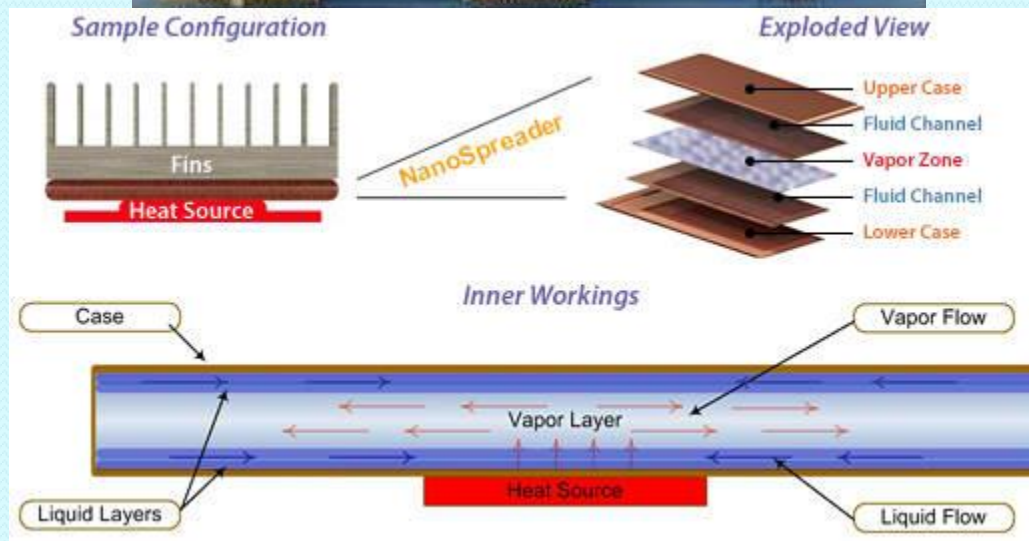


Temp Rise WO/W Heat Sink 15 Watts/Close to Radiator





Heat Pipe/NanoSpreader

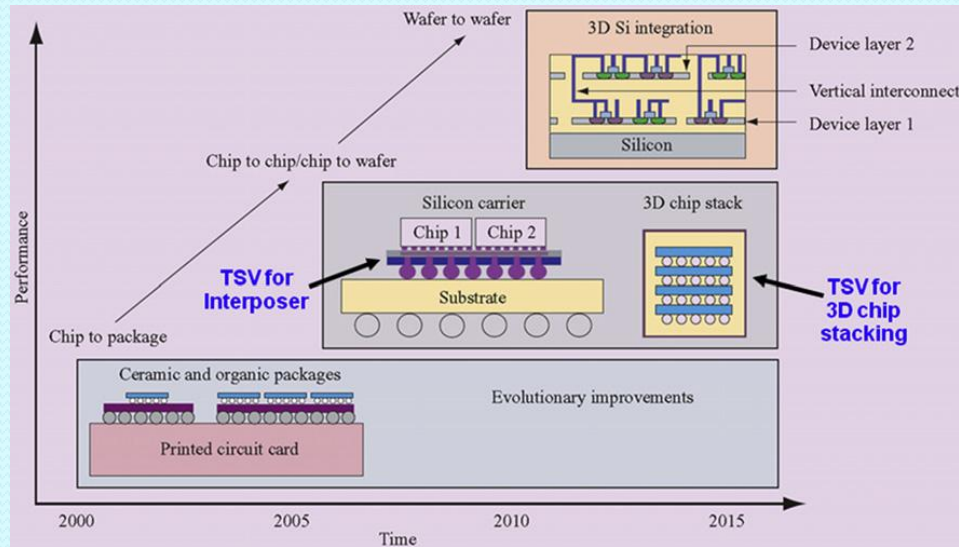


http://www.celsiatechnologies.com/nanospreader_technology.asp



TM for 3D ICs

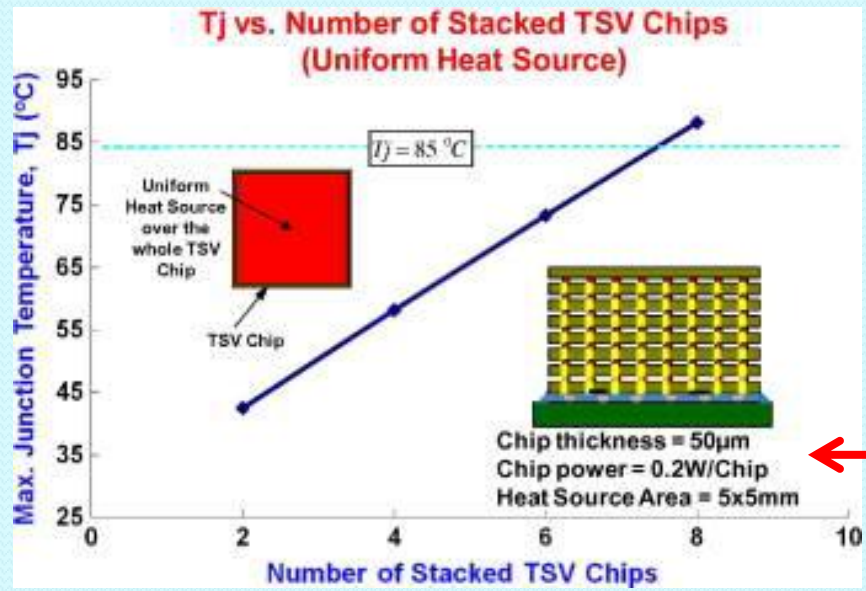
- Increase in power density
 - Heat of various stacks adds up
 - Cooling of inter-stacks is difficult
 - Heat flux different for each stack
- Local chip overheating
 - Inter-stack spacing is decreasing
 - Thinner die, hot spot
 - Partial remedy by TSV
- 3D stack hotter with
 - Higher stacks
 - Thinner die
 - Coarser TSV pitch, higher aspect ratio (t/d)
 - Thinner interposer, lower via conductivity
 - Smaller die



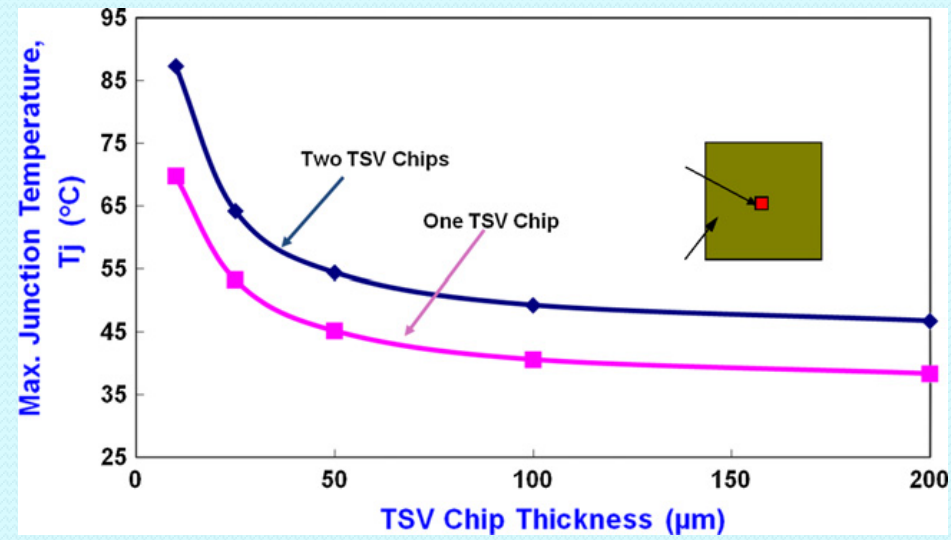
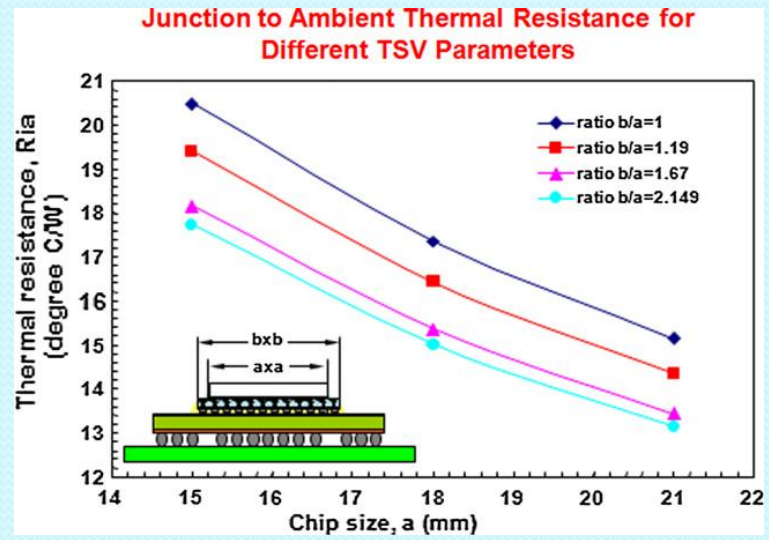
DRAM/ 0.4 W
5x5 mm/10 μm
Hot Spot
138°C >> 85 °C



3D IC & TSV/Die Effects on TM



**Chip Power
0.2 W/Chip
5x5 mm
50 micron**





Summary

TM Needs and Growth

- **Market**
 - Market projects study growth
 - High I/O FPGA/3D ICs continue to grow
 - Technology needs
- **Space Applications**
 - Limited on use of advanced technologies
 - Reliability and efficiency are critical
 - Processing power grows; therefore TM needs
- **Approaches**
 - Heat sink
 - Heat pipe
 - New tech. must address reliability /efficiency/reworkability
 - System approaches: hardware/software



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Visit
NEPP.NASA.gov